

Appendix C

The Finite Element Analysis of the Optimal Structures in Chapter 5

Six-bar truss with weight of 4962.1 lb (optimal result from the proposed ACO algorithm, Rajan, 1995 and Nanakorn and Meesomklin, 2001)

FINITE ELEMENT ANALYSIS PROGRAM
 C++ VERSION BY PRUETTHA NANAKORN
 1994.

VERSION : 960331

INPUT: 6bar4962.1noweight.txt

NOTE:

Number of nodal points = 5
 Number of elements = 6
 Number of material sets = 6
 Number of Dimensions = 2
 Number of Dof's / node (default) = 2
 Number of nodes / element (Maximum) = 2
 Number of load cases = 1

NODAL COORDINATES ::

NOTE:

NODE	1 COOR	2 COOR	dof
1	0.000000	0.000000	
2	360.000000	0.000000	
3	720.000000	0.000000	
4	0.000000	360.000000	
5	360.000000	360.000000	

ELEMENTS ::

NOTE:

ELEM	MAT	1 NODE	2 NODE
1	1	4	5
2	2	1	2
3	3	2	3
4	4	4	2
5	5	1	5
6	6	5	3

BOUNDARIES ::

NOTE:

NODE	1 NDF	2 NDF
1	1	1
4	1	1

NODAL FORCED BOUNDARY VALUE ::

NOTE:

LOAD CASE: 1	1 NDF	2 NDF
2	0.000000	-100.000000
3	0.000000	-100.000000

PROPORTIONAL LOADING ::

NOTE:

LOADING CONTROL

LOAD CASE: 1	Tmin A3	Tmax A4	A1 A5	A2 L
1	0.000000 0.000000	100.000000 0.000000	0.000000 0.000000	1.000000 0.000000

MATERIAL PROPERTIES ::

NOTE:

SET 1 Element Type 2

Young's Modulus =10000.000000
Area =30.000000

SET 2 Element Type 2
Young's Modulus =10000.000000
Area =19.900000

SET 3 Element Type 2
Young's Modulus =10000.000000
Area =15.500000

SET 4 Element Type 2
Young's Modulus =10000.000000
Area =7.220000

SET 5 Element Type 2
Young's Modulus =10000.000000
Area =22.000000

SET 6 Element Type 2
Young's Modulus =10000.000000
Area =22.000000

MACRO COMMANDS ::
NOTE:

Dt 1.000000
Loop 1.000000
Time
TangForm
SolvAddu
Disp 1.000000 1 1 5 1
Stre
Next
END

TIME= 1.000000
STEP= 1/1

DISPLACEMENT ::
USER FACTOR = 1.000000e+000
LOAD CASE 1 'S FACTOR = 1.000000e+000

NODE	1 COOR	2 COOR	1 NDF	2 NDF
1	0.000000	0.000000	0.000000e+000	0.000000e+000
2	360.000000	0.000000	-3.618090e-001	-1.772105e+000
3	720.000000	0.000000	-5.940671e-001	-1.999734e+000
4	0.000000	360.000000	0.000000e+000	0.000000e+000
5	360.000000	360.000000	2.400000e-001	-7.028335e-001

Stress 1 = 6.666667e+000
Stress 2 = -1.005025e+001
Stress 3 = -6.451613e+000
Stress 4 = 1.958745e+001
Stress 5 = -6.428243e+000
Stress 6 = 6.428243e+000
1 Max 0.000000e+000 6.666667e+000 3.000000e+001

Min 0.000000e+000
2 Max 0.000000e+000 -1.005025e+001 1.990000e+001
Min 2.000000e+000
3 Max 0.000000e+000 -6.451613e+000 1.550000e+001
Min 0.000000e+000
4 Max 1.000000e+000 1.958745e+001 7.220000e+000
Min 0.000000e+000
5 Max 0.000000e+000 -6.428243e+000 2.200000e+001
Min 0.000000e+000
6 Max 0.000000e+000 6.428243e+000 2.200000e+001
Min 0.000000e+000

Stressmax= 1.958745e+001

Stressmin= -1.005025e+001

AbsStressmax= 1.958745e+001

AbsStressmin= 1.005025e+001

NO CONVERGENCE TEST 1/1

Terminated normally by STOP.

Ten-bar truss with weight of 5490.7 lb (optimal result from the proposed ACO algorithm)

FINITE ELEMENT ANALYSIS PROGRAM
 C++ VERSION BY PRUETTHA NANAKORN
 1994.

VERSION : 960331

INPUT: 10bar5490.71Noweight.txt

NOTE:

Number of nodal points = 6
 Number of elements = 10
 Number of material sets = 10
 Number of Dimensions = 2
 Number of Dof's / node (default) = 2
 Number of nodes / element (Maximum) = 2
 Number of load cases = 1

NODAL COORDINATES ::

NOTE:

NODE	1 COOR	2 COOR	dof
1	0.000000	0.000000	
2	360.000000	0.000000	
3	720.000000	0.000000	
4	0.000000	360.000000	
5	360.000000	360.000000	
6	720.000000	360.000000	

ELEMENTS ::

NOTE:

ELEM	MAT	1 NODE	2 NODE
1	1	4	5
2	2	5	6
3	3	1	2
4	4	2	3
5	5	2	5
6	6	3	6
7	7	2	4
8	8	1	5
9	9	3	5
10	10	2	6

BOUNDARIES ::

NOTE:

NODE	1 NDF	2 NDF
1	1	1
4	1	1

NODAL FORCED BOUNDARY VALUE ::

NOTE:

LOAD CASE: 1

NODE	1 NDF	2 NDF
2	0.000000	-100.000000
3	0.000000	-100.000000

PROPORTIONAL LOADING ::

NOTE:

LOADING CONTROL

LOAD CASE: 1

	Tmin A3	Tmax A4	A1 A5	A2 L
1	0.000000 0.000000	100.000000 0.000000	0.000000 0.000000	1.000000 0.000000

MATERIAL PROPERTIES ::

NOTE:

SET 1 Element Type 2
Young's Modulus =10000.000000
Area =33.500000

SET 2 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 3 Element Type 2
Young's Modulus =10000.000000
Area =22.900000

SET 4 Element Type 2
Young's Modulus =10000.000000
Area =14.200000

SET 5 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 6 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 7 Element Type 2
Young's Modulus =10000.000000
Area =7.970000

SET 8 Element Type 2
Young's Modulus =10000.000000
Area =22.900000

SET 9 Element Type 2
Young's Modulus =10000.000000
Area =22.000000

SET 10 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

MACRO COMMANDS ::

NOTE:

Dt 1.000000
Loop 1.000000
Time
TangForm
SolvAddu
Disp 1.000000 1 1 6 1
Stre
Next
END

TIME= 1.000000
STEP= 1/1

DISPLACEMENT ::

USER FACTOR = 1.000000e+000
LOAD CASE 1 'S FACTOR = 1.000000e+000

NODE	1 COOR	2 COOR	1 NDF	2 NDF
1	0.000000	0.000000	0.000000e+000	0.000000e+000
2	360.000000	0.000000	-2.810740e-001	-1.287736e+000
3	720.000000	0.000000	-5.300487e-001	-1.998943e+000
4	0.000000	360.000000	0.000000e+000	0.000000e+000
5	360.000000	360.000000	2.377136e-001	-7.766470e-001
6	720.000000	360.000000	2.775648e-001	-1.959092e+000

Stress 1 = 6.603156e+000

Stress 2 = 1.106979e+000

Stress 3 = -7.807611e+000

Stress 4 = -6.915964e+000

Stress 5 = 1.419693e+001

Stress 6 = 1.106979e+000

Stress 7 = 1.398142e+001

Stress 8 = -7.485186e+000

Stress 9 = 6.312965e+000

Stress 10 = -1.565505e+000

1	Max 0.000000e+000	6.603156e+000	3.350000e+001
	Min 0.000000e+000		
2	Max 0.000000e+000	1.106979e+000	1.620000e+000
	Min 0.000000e+000		
3	Max 0.000000e+000	-7.807611e+000	2.290000e+001
	Min 2.000000e+000		
4	Max 0.000000e+000	-6.915964e+000	1.420000e+001
	Min 0.000000e+000		
5	Max 1.000000e+000	1.419693e+001	1.620000e+000
	Min 0.000000e+000		
6	Max 0.000000e+000	1.106979e+000	1.620000e+000
	Min 0.000000e+000		
7	Max 0.000000e+000	1.398142e+001	7.970000e+000
	Min 0.000000e+000		
8	Max 0.000000e+000	-7.485186e+000	2.290000e+001
	Min 0.000000e+000		
9	Max 0.000000e+000	6.312965e+000	2.200000e+001
	Min 0.000000e+000		
10	Max 0.000000e+000	-1.565505e+000	1.620000e+000
	Min 0.000000e+000		

Stressmax= 1.419693e+001

Stressmin= -7.807611e+000

AbsStressmax= 1.419693e+001

AbsStressmin= 7.807611e+000

NO CONVERGENCE TEST 1/1

Terminated normally by STOP.

Ten-bar truss with weight of 5458.3 lb (optimal result from Galante, 1996)

FINITE ELEMENT ANALYSIS PROGRAM
 C++ VERSION BY PRUETTHA NANAKORN
 1994.

VERSION : 960331

INPUT: 10barGalante5458.30noweight.txt
 NOTE:

Number of nodal points = 6
 Number of elements = 10
 Number of material sets = 10
 Number of Dimensions = 2
 Number of Dof's / node (default) = 2
 Number of nodes / element (Maximum) = 2
 Number of load cases = 1

NODAL COORDINATES ::
 NOTE:

NODE	1 COOR	2 COOR	dof
1	0.000000	0.000000	
2	360.000000	0.000000	
3	720.000000	0.000000	
4	0.000000	360.000000	
5	360.000000	360.000000	
6	720.000000	360.000000	

ELEMENTS ::
 NOTE:

ELEM	MAT	1 NODE	2 NODE
1	1	4	5
2	2	5	6
3	3	1	2
4	4	2	3
5	5	2	5
6	6	3	6
7	7	2	4
8	8	1	5
9	9	3	5
10	10	2	6

BOUNDARIES ::
 NOTE:

NODE	1 NDF	2 NDF
1	1	1
4	1	1

NODAL FORCED BOUNDARY VALUE ::
 NOTE:

LOAD CASE: 1		
NODE	1 NDF	2 NDF
2	0.000000	-100.000000
3	0.000000	-100.000000

PROPORTIONAL LOADING ::
 NOTE:

LOADING CONTROL

LOAD CASE: 1				
	Tmin A3	Tmax A4	A1 A5	A2 L
1	0.000000 0.000000	100.000000 0.000000	0.000000 0.000000	1.000000 0.000000

MATERIAL PROPERTIES ::

NOTE:

SET 1 Element Type 2
Young's Modulus =10000.000000
Area =33.500000

SET 2 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 3 Element Type 2
Young's Modulus =10000.000000
Area =22.000000

SET 4 Element Type 2
Young's Modulus =10000.000000
Area =14.200000

SET 5 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 6 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 7 Element Type 2
Young's Modulus =10000.000000
Area =7.970000

SET 8 Element Type 2
Young's Modulus =10000.000000
Area =22.900000

SET 9 Element Type 2
Young's Modulus =10000.000000
Area =22.000000

SET 10 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

MACRO COMMANDS ::
NOTE:

Dt	1.000000				
Loop	1.000000				
Time					
TangForm					
SolvAddU					
Disp	1.000000	1	1	6	1
Stre					
Next					
END					

TIME= 1.000000
STEP= 1/1

DISPLACEMENT ::
USER FACTOR = 1.000000e+000
LOAD CASE 1 'S FACTOR = 1.000000e+000

NODE	1 COOR	2 COOR	1 NDF	2 NDF
1	0.000000	0.000000	0.000000e+000	0.000000e+000
2	360.000000	0.000000	-2.920843e-001	-1.294935e+000
3	720.000000	0.000000	-5.411820e-001	-2.012268e+000
4	0.000000	360.000000	0.000000e+000	0.000000e+000
5	360.000000	360.000000	2.380342e-001	-7.782941e-001
6	720.000000	360.000000	2.768076e-001	-1.973495e+000

Stress 1 = 6.612061e+000

Stress 2 = 1.077039e+000

Stress 3 = -8.113452e+000

Stress 4 = -6.919380e+000

Stress 5 = 1.435114e+001

Stress 6 = 1.077039e+000

Stress 7 = 1.392849e+001

Stress 8 = -7.503610e+000

Stress 9 = 6.316083e+000

Stress 10 = -1.523163e+000

1	Max 0.000000e+000	6.612061e+000	3.350000e+001
	Min 0.000000e+000		
2	Max 0.000000e+000	1.077039e+000	1.620000e+000
	Min 0.000000e+000		
3	Max 0.000000e+000	-8.113452e+000	2.200000e+001
	Min 2.000000e+000		
4	Max 0.000000e+000	-6.919380e+000	1.420000e+001
	Min 0.000000e+000		
5	Max 1.000000e+000	1.435114e+001	1.620000e+000
	Min 0.000000e+000		
6	Max 0.000000e+000	1.077039e+000	1.620000e+000
	Min 0.000000e+000		
7	Max 0.000000e+000	1.392849e+001	7.970000e+000
	Min 0.000000e+000		
8	Max 0.000000e+000	-7.503610e+000	2.290000e+001
	Min 0.000000e+000		
9	Max 0.000000e+000	6.316083e+000	2.200000e+001
	Min 0.000000e+000		
10	Max 0.000000e+000	-1.523163e+000	1.620000e+000
	Min 0.000000e+000		

Stressmax= 1.435114e+001

Stressmin= -8.113452e+000

AbsStressmax= 1.435114e+001

AbsStressmin= 8.113452e+000

NO CONVERGENCE TEST 1/1

 Terminated normally by STOP.

Ten-bar truss with weight of 5499.4 lb (optimal result from Nanakorn and Meesomklin, 2001)

FINITE ELEMENT ANALYSIS PROGRAM
 C++ VERSION BY PRUETTHA NANAKORN
 1994.

VERSION : 960331

INPUT: 10barKonlakarn5499.4Noweight.txt
 NOTE:

Number of nodal points = 6
 Number of elements = 10
 Number of material sets = 10
 Number of Dimensions = 2
 Number of Dof's / node (default) = 2
 Number of nodes / element (Maximum) = 2
 Number of load cases = 1

NODAL COORDINATES ::
 NOTE:

NODE	1 COOR	2 COOR	dof
1	0.000000	0.000000	
2	360.000000	0.000000	
3	720.000000	0.000000	
4	0.000000	360.000000	
5	360.000000	360.000000	
6	720.000000	360.000000	

ELEMENTS ::
 NOTE:

ELEM	MAT	1 NODE	2 NODE
1	1	4	5
2	2	5	6
3	3	1	2
4	4	2	3
5	5	2	5
6	6	3	6
7	7	2	4
8	8	1	5
9	9	3	5
10	10	2	6

BOUNDARIES ::
 NOTE:

NODE	1 NDF	2 NDF
1	1	1
4	1	1

NODAL FORCED BOUNDARY VALUE ::
 NOTE:

LOAD CASE: 1	1 NDF	2 NDF
NODE		
2	0.000000	-100.000000
3	0.000000	-100.000000

PROPORTIONAL LOADING ::
 NOTE:

LOADING CONTROL

LOAD CASE: 1	Tmin A3	Tmax A4	A1 A5	A2 L
1	0.000000 0.000000	100.000000 0.000000	0.000000 0.000000	1.000000 0.000000

MATERIAL PROPERTIES ::
NOTE:

SET 1 Element Type 2
Young's Modulus =10000.000000
Area =33.500000

SET 2 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 3 Element Type 2
Young's Modulus =10000.000000
Area =22.900000

SET 4 Element Type 2
Young's Modulus =10000.000000
Area =15.500000

SET 5 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 6 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 7 Element Type 2
Young's Modulus =10000.000000
Area =7.220000

SET 8 Element Type 2
Young's Modulus =10000.000000
Area =22.900000

SET 9 Element Type 2
Young's Modulus =10000.000000
Area =22.000000

SET 10 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

MACRO COMMANDS ::
NOTE:

Dt	1.000000				
Loop	1.000000				
Time					
TangForm					
SolvAddu					
Disp	1.000000	1	1	6	1
Stre					
Next					
END					

TIME= 1.000000
STEP= 1/1

DISPLACEMENT ::

USER FACTOR = 1.000000e+000
LOAD CASE 1 'S FACTOR = 1.000000e+000

NODE	1 COOR	2 COOR	1 NDF	2 NDF
1	0.000000	0.000000	0.000000e+000	0.000000e+000
2	360.000000	0.000000	-2.768093e-001	-1.349783e+000
3	720.000000	0.000000	-5.062836e-001	-1.995824e+000
4	0.000000	360.000000	0.000000e+000	0.000000e+000
5	360.000000	360.000000	2.406289e-001	-7.916248e-001
6	720.000000	360.000000	2.672634e-001	-1.969189e+000

Stress 1 = 6.684136e+000

Stress 2 = 7.398462e-001

Stress 3 = -7.689146e+000

Stress 4 = -6.374287e+000

Stress 5 = 1.550439e+001

Stress 6 = 7.398462e-001

Stress 7 = 1.490241e+001

Stress 8 = -7.652721e+000

Stress 9 = 6.351198e+000

Stress 10 = -1.046300e+000

1	Max 0.000000e+000	6.684136e+000	3.350000e+001
	Min 0.000000e+000		
2	Max 0.000000e+000	7.398462e-001	1.620000e+000
	Min 0.000000e+000		
3	Max 0.000000e+000	-7.689146e+000	2.290000e+001
	Min 2.000000e+000		
4	Max 0.000000e+000	-6.374287e+000	1.550000e+001
	Min 0.000000e+000		
5	Max 1.000000e+000	1.550439e+001	1.620000e+000
	Min 0.000000e+000		
6	Max 0.000000e+000	7.398462e-001	1.620000e+000
	Min 0.000000e+000		
7	Max 0.000000e+000	1.490241e+001	7.220000e+000
	Min 0.000000e+000		
8	Max 0.000000e+000	-7.652721e+000	2.290000e+001
	Min 0.000000e+000		
9	Max 0.000000e+000	6.351198e+000	2.200000e+001
	Min 0.000000e+000		
10	Max 0.000000e+000	-1.046300e+000	1.620000e+000
	Min 0.000000e+000		

Stressmax= 1.550439e+001

Stressmin= -7.689146e+000

AbsStressmax= 1.550439e+001

AbsStressmin= 7.689146e+000

NO CONVERGENCE TEST 1/1

Terminated normally by STOP.

Ten-bar truss with weight of 5613.6 lb (optimal result from Rajeev and Krishnamoorthy, 1992)

FINITE ELEMENT ANALYSIS PROGRAM
 C++ VERSION BY PRUETTHA NANAKORN
 1994.

VERSION : 960331

INPUT: 10barRajeev5613.6NoWeight.txt
 NOTE:

Number of nodal points = 6
 Number of elements = 10
 Number of material sets = 10
 Number of Dimensions = 2
 Number of Dof's / node (default) = 2
 Number of nodes / element (Maximum) = 2
 Number of load cases = 1

NODAL COORDINATES ::
 NOTE:

NODE	1 COOR	2 COOR	dof
1	0.000000	0.000000	
2	360.000000	0.000000	
3	720.000000	0.000000	
4	0.000000	360.000000	
5	360.000000	360.000000	
6	720.000000	360.000000	

ELEMENTS ::
 NOTE:

ELEM	MAT	1 NODE	2 NODE
1	1	4	5
2	2	5	6
3	3	1	2
4	4	2	3
5	5	2	5
6	6	3	6
7	7	2	4
8	8	1	5
9	9	3	5
10	10	2	6

BOUNDARIES ::
 NOTE:

NODE	1 NDF	2 NDF
1	1	1
4	1	1

NODAL FORCED BOUNDARY VALUE ::
 NOTE:

LOAD CASE: 1

NODE	1 NDF	2 NDF
2	0.000000	-100.000000
3	0.000000	-100.000000

PROPORTIONAL LOADING ::
 NOTE:

LOADING CONTROL

LOAD CASE: 1

	Tmin A3	Tmax A4	A1 A5	A2 L
1	0.000000 0.000000	100.000000 0.000000	0.000000 0.000000	1.000000 0.000000

MATERIAL PROPERTIES ::
NOTE:

SET 1 Element Type 2
Young's Modulus =10000.000000
Area =33.500000

SET 2 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 3 Element Type 2
Young's Modulus =10000.000000
Area =22.000000

SET 4 Element Type 2
Young's Modulus =10000.000000
Area =15.500000

SET 5 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 6 Element Type 2
Young's Modulus =10000.000000
Area =1.620000

SET 7 Element Type 2
Young's Modulus =10000.000000
Area =14.200000

SET 8 Element Type 2
Young's Modulus =10000.000000
Area =19.900000

SET 9 Element Type 2
Young's Modulus =10000.000000
Area =19.900000

SET 10 Element Type 2
Young's Modulus =10000.000000
Area =2.620000

MACRO COMMANDS ::
NOTE:

Dt	1.000000				
Loop	1.000000				
Time					
TangForm					
SolvAddU					
Disp	1.000000	1	1	6	1
Stre					
Next					
END					

TIME= 1.000000
STEP= 1/1

DISPLACEMENT ::

USER FACTOR = 1.000000e+000
LOAD CASE 1 'S FACTOR = 1.000000e+000

NODE	1 COOR	2 COOR	1 NDF	2 NDF
1	0.000000	0.000000	0.000000e+000	0.000000e+000
2	360.000000	0.000000	-3.187346e-001	-9.983859e-001
3	720.000000	0.000000	-5.380818e-001	-2.000752e+000
4	0.000000	360.000000	0.000000e+000	0.000000e+000
5	360.000000	360.000000	2.205325e-001	-7.589057e-001
6	720.000000	360.000000	3.440626e-001	-1.877222e+000

Stress 1 = 6.125903e+000

Stress 2 = 3.431392e+000

Stress 3 = -8.853739e+000

Stress 4 = -6.092977e+000

Stress 5 = 6.652227e+000

Stress 6 = 3.431392e+000

Stress 7 = 9.439601e+000

Stress 8 = -7.477406e+000

Stress 9 = 6.711555e+000

Stress 10 = -3.000538e+000

1	Max 0.000000e+000	6.125903e+000	3.350000e+001
	Min 0.000000e+000		
2	Max 0.000000e+000	3.431392e+000	1.620000e+000
	Min 0.000000e+000		
3	Max 0.000000e+000	-8.853739e+000	2.200000e+001
	Min 2.000000e+000		
4	Max 0.000000e+000	-6.092977e+000	1.550000e+001
	Min 0.000000e+000		
5	Max 0.000000e+000	6.652227e+000	1.620000e+000
	Min 0.000000e+000		
6	Max 0.000000e+000	3.431392e+000	1.620000e+000
	Min 0.000000e+000		
7	Max 1.000000e+000	9.439601e+000	1.420000e+001
	Min 0.000000e+000		
8	Max 0.000000e+000	-7.477406e+000	1.990000e+001
	Min 0.000000e+000		
9	Max 0.000000e+000	6.711555e+000	1.990000e+001
	Min 0.000000e+000		
10	Max 0.000000e+000	-3.000538e+000	2.620000e+000
	Min 0.000000e+000		

Stressmax= 9.439601e+000

Stressmin= -8.853739e+000

AbsStressmax= 9.439601e+000

AbsStressmin= 8.853739e+000

NO CONVERGENCE TEST 1/1

Terminated normally by STOP.

Fifty-two-bar truss with weight of 1902.6 kgf (optimal result from the proposed ACO algorithm)

FINITE ELEMENT ANALYSIS PROGRAM
 C++ VERSION BY PRUETTHA NANAKORN
 1994.

VERSION : 960331

INPUT: 52bar1902.61Noweight.txt

NOTE:

Number of nodal points = 20
 Number of elements = 52
 Number of material sets = 12
 Number of Dimensions = 2
 Number of Dof's / node (default) = 2
 Number of nodes / element (Maximum) = 2
 Number of load cases = 1

NODAL COORDINATES ::

NOTE:

NODE	1 COOR	2 COOR	dof
1	0.000000	0.000000	
2	2000.000000	0.000000	
3	4000.000000	0.000000	
4	6000.000000	0.000000	
5	0.000000	3000.000000	
6	2000.000000	3000.000000	
7	4000.000000	3000.000000	
8	6000.000000	3000.000000	
9	0.000000	6000.000000	
10	2000.000000	6000.000000	
11	4000.000000	6000.000000	
12	6000.000000	6000.000000	
13	0.000000	9000.000000	
14	2000.000000	9000.000000	
15	4000.000000	9000.000000	
16	6000.000000	9000.000000	
17	0.000000	12000.000000	
18	2000.000000	12000.000000	
19	4000.000000	12000.000000	
20	6000.000000	12000.000000	

ELEMENTS ::

NOTE:

ELEM	MAT	1 NODE	2 NODE
1	1	1	5
2	1	2	6
3	1	3	7
4	1	4	8
5	2	2	5
6	2	1	6
7	2	3	6
8	2	2	7
9	2	4	7
10	2	3	8
11	3	5	6
12	3	6	7
13	3	7	8
14	4	5	9
15	4	6	10
16	4	7	11
17	4	8	12
18	5	6	9
19	5	5	10
20	5	7	10
21	5	6	11
22	5	8	11
23	5	7	12
24	6	9	10
25	6	10	11

26	6	11	12
27	7	9	13
28	7	10	14
29	7	11	15
30	7	12	16
31	8	10	13
32	8	9	14
33	8	11	14
34	8	10	15
35	8	12	15
36	8	11	16
37	9	13	14
38	9	14	15
39	9	15	16
40	10	13	17
41	10	14	18
42	10	15	19
43	10	16	20
44	11	14	17
45	11	13	18
46	11	15	18
47	11	14	19
48	11	16	19
49	11	15	20
50	12	17	18
51	12	18	19
52	12	19	20

BOUNDARIES ::

NOTE:

NODE	1 NDF	2 NDF
1	1	1
2	1	1
3	1	1
4	1	1

NODAL FORCED BOUNDARY VALUE ::

NOTE:

LOAD CASE: 1		
NODE	1 NDF	2 NDF
17	100000.000000	200000.000000
18	100000.000000	200000.000000
19	100000.000000	200000.000000
20	100000.000000	200000.000000

PROPORTIONAL LOADING ::

NOTE:

LOADING CONTROL

LOAD CASE: 1				
	Tmin A3	Tmax A4	A1 A5	A2 L
1	0.000000 0.000000	100.000000 0.000000	0.000000 0.000000	1.000000 0.000000

MATERIAL PROPERTIES ::

NOTE:

SET 1 Element Type 2
 Young's Modulus =207000.000000
 Area =4658.055000

SET 2 Element Type 2
 Young's Modulus =207000.000000
 Area =1161.288000

SET 3 Element Type 2
 Young's Modulus =207000.000000
 Area =494.193000

SET 4 Element Type 2
Young's Modulus =207000.000000
Area =3303.219000

SET 5 Element Type 2
Young's Modulus =207000.000000
Area =940.000000

SET 6 Element Type 2
Young's Modulus =207000.000000
Area =494.193000

SET 7 Element Type 2
Young's Modulus =207000.000000
Area =2238.705000

SET 8 Element Type 2
Young's Modulus =207000.000000
Area =1008.385000

SET 9 Element Type 2
Young's Modulus =207000.000000
Area =494.193000

SET 10 Element Type 2
Young's Modulus =207000.000000
Area =1283.868000

SET 11 Element Type 2
Young's Modulus =207000.000000
Area =1161.288000

SET 12 Element Type 2
Young's Modulus =207000.000000
Area =494.193000

MACRO COMMANDS ::
NOTE:
Dt 1.000000
Loop 1.000000
Time
TangForm
SolvAddu
Disp 1.000000 1 1 20 1
Stre
Next
END

TIME= 1.000000
STEP= 1/1

DISPLACEMENT ::
USER FACTOR = 1.000000e+000
LOAD CASE 1 'S FACTOR = 1.000000e+000

NODE	1 COOR	2 COOR	1 NDF	2 NDF
1	0.000000	0.000000	0.000000e+000	0.000000e+000

2	2000.000000	0.000000	0.000000e+000	0.000000e+000
3	4000.000000	0.000000	0.000000e+000	0.000000e+000
4	6000.000000	0.000000	0.000000e+000	0.000000e+000
5	0.000000	3000.000000	5.240107e+000	2.431419e+000
6	2000.000000	3000.000000	4.071861e+000	9.389795e-001
7	4000.000000	3000.000000	3.766583e+000	6.209381e-002
8	6000.000000	3000.000000	4.166531e+000	-1.291827e+000
9	0.000000	6000.000000	1.223166e+001	5.036712e+000
10	2000.000000	6000.000000	1.104212e+001	2.274434e+000
11	4000.000000	6000.000000	1.047975e+001	2.600235e-001
12	6000.000000	6000.000000	1.060255e+001	-2.377064e+000
13	0.000000	9000.000000	2.060267e+001	7.640765e+000
14	2000.000000	9000.000000	1.937449e+001	3.803048e+000
15	4000.000000	9000.000000	1.875106e+001	8.456983e-001
16	6000.000000	9000.000000	1.840034e+001	-2.892713e+000
17	0.000000	12000.000000	2.986551e+001	1.020626e+001
18	2000.000000	12000.000000	2.826587e+001	5.448211e+000
19	4000.000000	12000.000000	2.751277e+001	1.917690e+000
20	6000.000000	12000.000000	2.769513e+001	-2.170358e+000

Stressmax= 1.797652e+002

Stressmin= -1.655633e+002

AbsStressmax= 1.797652e+002

AbsStressmin= 1.655633e+002
NO CONVERGENCE TEST 1/1

Terminated normally by STOP.

Fifty-two-bar truss with weight of 1972.7 kgf (optimal result from Wu and Chow, 1995)

FINITE ELEMENT ANALYSIS PROGRAM
 C++ VERSION BY PRUETTHA NANAKORN
 1994.

VERSION : 960331

INPUT: 52bar1972.7shyueJianWuNoWeight.txt
 NOTE:

Number of nodal points = 20
 Number of elements = 52
 Number of material sets = 12
 Number of Dimensions = 2
 Number of Dof's / node (default) = 2
 Number of nodes / element (Maximum) = 2
 Number of load cases = 1

NODAL COORDINATES ::
 NOTE:

NODE	1 COOR	2 COOR	dof
1	0.000000	0.000000	
2	2000.000000	0.000000	
3	4000.000000	0.000000	
4	6000.000000	0.000000	
5	0.000000	3000.000000	
6	2000.000000	3000.000000	
7	4000.000000	3000.000000	
8	6000.000000	3000.000000	
9	0.000000	6000.000000	
10	2000.000000	6000.000000	
11	4000.000000	6000.000000	
12	6000.000000	6000.000000	
13	0.000000	9000.000000	
14	2000.000000	9000.000000	
15	4000.000000	9000.000000	
16	6000.000000	9000.000000	
17	0.000000	12000.000000	
18	2000.000000	12000.000000	
19	4000.000000	12000.000000	
20	6000.000000	12000.000000	

ELEMENTS ::
 NOTE:

ELEM	MAT	1 NODE	2 NODE
1	1	1	5
2	1	2	6
3	1	3	7
4	1	4	8
5	2	2	5
6	2	1	6
7	2	3	6
8	2	2	7
9	2	4	7
10	2	3	8
11	3	5	6
12	3	6	7
13	3	7	8
14	4	5	9
15	4	6	10
16	4	7	11
17	4	8	12
18	5	6	9
19	5	5	10
20	5	7	10
21	5	6	11
22	5	8	11
23	5	7	12
24	6	9	10
25	6	10	11

26	6	11	12
27	7	9	13
28	7	10	14
29	7	11	15
30	7	12	16
31	8	10	13
32	8	9	14
33	8	11	14
34	8	10	15
35	8	12	15
36	8	11	16
37	9	13	14
38	9	14	15
39	9	15	16
40	10	13	17
41	10	14	18
42	10	15	19
43	10	16	20
44	11	14	17
45	11	13	18
46	11	15	18
47	11	14	19
48	11	16	19
49	11	15	20
50	12	17	18
51	12	18	19
52	12	19	20

BOUNDARIES ::
NOTE:

NODE	1 NDF	2 NDF
1	1	1
2	1	1
3	1	1
4	1	1

NODAL FORCED BOUNDARY VALUE ::
NOTE:

LOAD CASE: 1	1 NDF	2 NDF
17	100000.000000	200000.000000
18	100000.000000	200000.000000
19	100000.000000	200000.000000
20	100000.000000	200000.000000

PROPORTIONAL LOADING ::
NOTE:

LOADING CONTROL

LOAD CASE: 1	Tmin A3	Tmax A4	A1 A5	A2 L
1	0.000000 0.000000	100.000000 0.000000	0.000000 0.000000	1.000000 0.000000

MATERIAL PROPERTIES ::
NOTE:

SET 1 Element Type 2
Young's Modulus =207000.000000
Area =4658.055000

SET 2 Element Type 2
Young's Modulus =207000.000000
Area =1161.288000

SET 3 Element Type 2
Young's Modulus =207000.000000
Area =645.160000

SET 4 Element Type 2
Young's Modulus =207000.000000
Area =3303.219000

SET 5 Element Type 2
Young's Modulus =207000.000000
Area =1045.159000

SET 6 Element Type 2
Young's Modulus =207000.000000
Area =494.193000

SET 7 Element Type 2
Young's Modulus =207000.000000
Area =2477.414000

SET 8 Element Type 2
Young's Modulus =207000.000000
Area =1045.159000

SET 9 Element Type 2
Young's Modulus =207000.000000
Area =285.161000

SET 10 Element Type 2
Young's Modulus =207000.000000
Area =1696.771000

SET 11 Element Type 2
Young's Modulus =207000.000000
Area =1045.159000

SET 12 Element Type 2
Young's Modulus =207000.000000
Area =641.289000

MACRO COMMANDS ::
NOTE:

Dt 1.000000
Loop 1.000000
Time
TangForm
SolvAddu
Disp 1.000000 1 1 20 1
Stre
Next
END

TIME= 1.000000
STEP= 1/1

DISPLACEMENT ::
USER FACTOR = 1.000000e+000
LOAD CASE 1 'S FACTOR = 1.000000e+000

NODE	1 COOR	2 COOR	1 NDF	2 NDF
1	0.000000	0.000000	0.000000e+000	0.000000e+000

2	2000.000000	0.000000	0.000000e+000	0.000000e+000
3	4000.000000	0.000000	0.000000e+000	0.000000e+000
4	6000.000000	0.000000	0.000000e+000	0.000000e+000
5	0.000000	3000.000000	5.101229e+000	2.422932e+000
6	2000.000000	3000.000000	4.100765e+000	9.375873e-001
7	4000.000000	3000.000000	3.812780e+000	5.650918e-002
8	6000.000000	3000.000000	4.130940e+000	-1.284134e+000
9	0.000000	6000.000000	1.181564e+001	5.003919e+000
10	2000.000000	6000.000000	1.062700e+001	2.237866e+000
11	4000.000000	6000.000000	1.004265e+001	2.567451e-001
12	6000.000000	6000.000000	1.017573e+001	-2.359833e+000
13	0.000000	9000.000000	2.017897e+001	7.388327e+000
14	2000.000000	9000.000000	1.869967e+001	3.686808e+000
15	4000.000000	9000.000000	1.804816e+001	8.170591e-001
16	6000.000000	9000.000000	1.771993e+001	-2.847064e+000
17	0.000000	12000.000000	2.881764e+001	9.446366e+000
18	2000.000000	12000.000000	2.772231e+001	5.153455e+000
19	4000.000000	12000.000000	2.716783e+001	1.786952e+000
20	6000.000000	12000.000000	2.736410e+001	-2.253092e+000

Stressmax= 1.789236e+002

Stressmin= -1.531071e+002

AbsStressmax= 1.789236e+002

AbsStressmin= 1.531071e+002
NO CONVERGENCE TEST 1/1

Terminated normally by STOP.

One-bay eight-story frame with weight of 7032.4 lb (optimal result from the proposed ACO algorithm)

FINITE ELEMENT ANALYSIS PROGRAM
 C++ VERSION BY PRUETTHA NANAKORN
 1994.

VERSION : 20021101

Number of nodal points = 18
 Number of elements = 24
 Number of material sets = 8
 Number of Dimensions = 2
 Number of Dof's / node (default) = 3
 Number of nodes / element (Maximum) = 2
 Number of load cases = 1

NODAL COORDINATES ::
 NOTE:

NODE	1 COOR	2 COOR	dof
1	0.000000	0.000000	
2	120.000000	0.000000	
3	0.000000	120.000000	
4	120.000000	120.000000	
5	0.000000	240.000000	
6	120.000000	240.000000	
7	0.000000	360.000000	
8	120.000000	360.000000	
9	0.000000	480.000000	
10	120.000000	480.000000	
11	0.000000	600.000000	
12	120.000000	600.000000	
13	0.000000	720.000000	
14	120.000000	720.000000	
15	0.000000	840.000000	
16	120.000000	840.000000	
17	0.000000	960.000000	
18	120.000000	960.000000	

ELEMENTS ::
 NOTE:

ELEM	MAT	1 NODE	2 NODE
1	1	1	3
2	1	2	4
3	1	3	5
4	1	4	6
5	2	5	7
6	2	6	8
7	2	7	9
8	2	8	10
9	3	9	11
10	3	10	12
11	3	11	13
12	3	12	14
13	4	13	15
14	4	14	16
15	4	15	17
16	4	16	18
17	5	3	4
18	5	5	6
19	6	7	8
20	6	9	10
21	7	11	12
22	7	13	14
23	8	15	16
24	8	17	18

BOUNDARIES ::
 NOTE:

NODE	1 NDF	2 NDF	3 NDF
1	1	1	1
2	1	1	1

NODAL FORCED BOUNDARY VALUE ::
NOTE:

LOAD CASE: 1	1 NDF	2 NDF	3 NDF
NODE			
1	0.000000	-100.000000	0.000000
2	0.000000	-100.000000	0.000000
3	0.272000	-100.000000	0.000000
4	0.000000	-100.000000	0.000000
5	0.544000	-100.000000	0.000000
6	0.000000	-100.000000	0.000000
7	0.816000	-100.000000	0.000000
8	0.000000	-100.000000	0.000000
9	1.088000	-100.000000	0.000000
10	0.000000	-100.000000	0.000000
11	1.361000	-100.000000	0.000000
12	0.000000	-100.000000	0.000000
13	1.633000	-100.000000	0.000000
14	0.000000	-100.000000	0.000000
15	1.905000	-100.000000	0.000000
16	0.000000	-100.000000	0.000000
17	2.831000	-100.000000	0.000000
18	0.000000	-100.000000	0.000000

PROPORTIONAL LOADING ::
NOTE:

LOADING CONTROL

LOAD CASE: 1	Tmin A3	Tmax A4	A1 A5	A2 L
1	0.000000 0.000000	100.000000 0.000000	1.000000 0.000000	0.000000 0.000000

MATERIAL PROPERTIES ::
NOTE:

SET 1 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 10.300000
Izz = 510.000000
Hinge Data:
Number of hinges = 0

Distribute loads
QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 2 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 10.300000
Izz = 510.000000
Hinge Data:
Number of hinges = 0

Distribute loads
QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 3 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 7.680000
Izz = 301.000000
Hinge Data:

Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 4 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 4.160000
IZZ = 88.600000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 5 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 10.300000
IZZ = 510.000000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 6 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 13.000000
IZZ = 843.000000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 7 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 7.680000
IZZ = 301.000000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 8 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 7.680000
IZZ = 301.000000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

MACRO COMMANDS ::
NOTE:

```

Dt          1.000000
Loop        1.000000
Time
Tang
Form
Solv
Stre
FacU
AddU
SumUserFac
Disp       1.000000      1      1      18      1
Next
END

```

```

-----
-----
TIME= 1.000000
STEP= 1/1

```

```

DISPLACEMENT ::
USER FACTOR = 1.000000e+000
LOAD CASE 1 'S FACTOR = 1.000000e+000

```

NODE	1 COOR	2 COOR	1 NDF	2 NDF	3 NDF
1	0.000000	0.000000	0.000000e+000	-1.000000e+002	0.000000e+000
2	120.000000	0.000000	0.000000e+000	-1.000000e+002	0.000000e+000
3	0.000000	120.000000	1.151100e-001	-1.003000e+002	-1.070424e-003
4	120.000000	120.000000	1.150557e-001	-1.003428e+002	-1.069930e-003
5	0.000000	240.000000	3.127886e-001	-1.005633e+002	-1.398336e-003
6	120.000000	240.000000	3.126796e-001	-1.006419e+002	-1.398083e-003
7	0.000000	360.000000	5.253018e-001	-1.007901e+002	-1.361904e-003
8	120.000000	360.000000	5.251720e-001	-1.008972e+002	-1.361776e-003
9	0.000000	480.000000	7.395457e-001	-1.009806e+002	-1.493389e-003
10	120.000000	480.000000	7.393717e-001	-1.011084e+002	-1.492743e-003
11	0.000000	600.000000	1.018161e+000	-1.011869e+002	-2.086835e-003
12	120.000000	600.000000	1.017796e+000	-1.013331e+002	-2.085689e-003
13	0.000000	720.000000	1.320277e+000	-1.013428e+002	-2.072958e-003
14	120.000000	720.000000	1.319837e+000	-1.015006e+002	-2.072679e-003
15	0.000000	840.000000	1.692363e+000	-1.015365e+002	-1.916575e-003
16	120.000000	840.000000	1.691849e+000	-1.017047e+002	-1.915658e-003
17	0.000000	960.000000	1.984917e+000	-1.016345e+002	-1.636038e-003
18	120.000000	960.000000	1.984155e+000	-1.018057e+002	-1.635059e-003

```

NO CONVERGENCE TEST 1/1
-----
-----

```

```

Terminated normally by STOP.
-----

```

One-bay eight-story frame with weight of 7376.1 lb (optimal result from Camp et al., 1998)

FINITE ELEMENT ANALYSIS PROGRAM
 C++ VERSION BY PRUETTHA NANAKORN
 1994.

VERSION : 20021101

Number of nodal points = 18
 Number of elements = 24
 Number of material sets = 8
 Number of Dimensions = 2
 Number of Dof's / node (default) = 3
 Number of nodes / element (Maximum) = 2
 Number of load cases = 1

NODAL COORDINATES ::
 NOTE:

NODE	1 COOR	2 COOR	dof
1	0.000000	0.000000	
2	120.000000	0.000000	
3	0.000000	120.000000	
4	120.000000	120.000000	
5	0.000000	240.000000	
6	120.000000	240.000000	
7	0.000000	360.000000	
8	120.000000	360.000000	
9	0.000000	480.000000	
10	120.000000	480.000000	
11	0.000000	600.000000	
12	120.000000	600.000000	
13	0.000000	720.000000	
14	120.000000	720.000000	
15	0.000000	840.000000	
16	120.000000	840.000000	
17	0.000000	960.000000	
18	120.000000	960.000000	

ELEMENTS ::
 NOTE:

ELEM	MAT	1 NODE	2 NODE
1	1	1	3
2	1	2	4
3	1	3	5
4	1	4	6
5	2	5	7
6	2	6	8
7	2	7	9
8	2	8	10
9	3	9	11
10	3	10	12
11	3	11	13
12	3	12	14
13	4	13	15
14	4	14	16
15	4	15	17
16	4	16	18
17	5	3	4
18	5	5	6
19	6	7	8
20	6	9	10
21	7	11	12
22	7	13	14
23	8	15	16
24	8	17	18

BOUNDARIES ::
 NOTE:

NODE	1 NDF	2 NDF	3 NDF
1	1	1	1
2	1	1	1

NODAL FORCED BOUNDARY VALUE ::
NOTE:

LOAD CASE: 1

NODE	1 NDF	2 NDF	3 NDF
1	0.000000	-100.000000	0.000000
2	0.000000	-100.000000	0.000000
3	0.272000	-100.000000	0.000000
4	0.000000	-100.000000	0.000000
5	0.544000	-100.000000	0.000000
6	0.000000	-100.000000	0.000000
7	0.816000	-100.000000	0.000000
8	0.000000	-100.000000	0.000000
9	1.088000	-100.000000	0.000000
10	0.000000	-100.000000	0.000000
11	1.361000	-100.000000	0.000000
12	0.000000	-100.000000	0.000000
13	1.633000	-100.000000	0.000000
14	0.000000	-100.000000	0.000000
15	1.905000	-100.000000	0.000000
16	0.000000	-100.000000	0.000000
17	2.831000	-100.000000	0.000000
18	0.000000	-100.000000	0.000000

PROPORTIONAL LOADING ::
NOTE:

LOADING CONTROL

LOAD CASE: 1

	Tmin A3	Tmax A4	A1 A5	A2 L
1	0.000000 0.000000	100.000000 0.000000	1.000000 0.000000	0.000000 0.000000

MATERIAL PROPERTIES ::
NOTE:

SET 1 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 13.500000
Izz = 712.000000
Hinge Data:
Number of hinges = 0

Distribute loads
QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 2 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 9.120000
Izz = 375.000000
Hinge Data:
Number of hinges = 0

Distribute loads
QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 3 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 7.680000
Izz = 301.000000
Hinge Data:

Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 4 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 4.710000
Izz = 103.000000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 5 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 10.300000
Izz = 510.000000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 6 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 10.300000
Izz = 510.000000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 7 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 10.300000
Izz = 510.000000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

SET 8 Element Type 4
Two-Noded Two-Dimensional Beam-Column Element with Semi-Rigid Ends.
Young's Modulus = 29000.000000
Area = 7.680000
Izz = 301.000000

Hinge Data:
Number of hinges = 0

Distribute loads

QaA0 = 0.000000
QbA0 = 0.000000
QbA1 = 0.000000

MACRO COMMANDS ::
NOTE:

```

Dt          1.000000
Loop        1.000000
Time
Tang
Form
Solv
Stre
FacU
AddU
SumUserFac
Disp       1.000000      1      1      18      1
Next
END

```

```

-----
-----
TIME= 1.000000
STEP= 1/1

```

```

DISPLACEMENT ::
USER FACTOR = 1.000000e+000
LOAD CASE 1 'S FACTOR = 1.000000e+000

```

NODE	1 COOR	2 COOR	1 NDF	2 NDF	3 NDF
1	0.000000	0.000000	0.000000e+000	-1.000000e+002	0.000000e+000
2	120.000000	0.000000	0.000000e+000	-1.000000e+002	0.000000e+000
3	0.000000	120.000000	9.375979e-002	-1.002290e+002	-9.551405e-004
4	120.000000	120.000000	9.370577e-002	-1.002614e+002	-9.546400e-004
5	0.000000	240.000000	2.620693e-001	-1.004300e+002	-1.258478e-003
6	120.000000	240.000000	2.619599e-001	-1.004895e+002	-1.258105e-003
7	0.000000	360.000000	4.910806e-001	-1.006864e+002	-1.495233e-003
8	120.000000	360.000000	4.909167e-001	-1.007776e+002	-1.494861e-003
9	0.000000	480.000000	7.362968e-001	-1.009015e+002	-1.618598e-003
10	120.000000	480.000000	7.360782e-001	-1.010162e+002	-1.618230e-003
11	0.000000	600.000000	9.979905e-001	-1.011075e+002	-1.680193e-003
12	120.000000	600.000000	9.977172e-001	-1.012413e+002	-1.679863e-003
13	0.000000	720.000000	1.251416e+000	-1.012633e+002	-1.667951e-003
14	120.000000	720.000000	1.251088e+000	-1.014087e+002	-1.667457e-003
15	0.000000	840.000000	1.572868e+000	-1.014345e+002	-1.786325e-003
16	120.000000	840.000000	1.572355e+000	-1.015890e+002	-1.784972e-003
17	0.000000	960.000000	1.839543e+000	-1.015210e+002	-1.520030e-003
18	120.000000	960.000000	1.838781e+000	-1.016782e+002	-1.519042e-003

```

NO CONVERGENCE TEST 1/1
-----
-----

```

```

Terminated normally by STOP.
-----

```